Lecture 4: physical models in mechanical and aeronautic engineering, numerical solutions, formulation of optimal design problems

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1. The membrane's equation, reminders on FEM and weak formulations

- 2. Linear elasticity
- 3. Fluid flows
- 4. Heat diffusion

5. Coupled physics:

- 5.1 convective heat transfer
- 5.2 fluid-structure interactions
- 5.3 thermoelasticity

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A non-exhaustive review of shape and topology optimization techniques Shape optimization problems

Shape/Topology optimization is the mathematical art of generating shapes that best fulfill a proposed objective.

Generically, a design optimization problem arises under the form

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m ad}}J(\Omega)\ &s.t. egin{cases} G_i(\Omega)=0, & 1\leq i\leq p\ H_j(\Omega)\leq 0, & 1\leq j\leq q \end{aligned}$$

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In industrial applications, $J(\Omega)$, $G_i(\Omega)$ or $H_j(\Omega)$ involve the solution u_{Ω} defined with respect to a PDE model posed on Ω .





$$\begin{cases} -\text{div}(A\nabla u)=f \text{ in }\Omega,\\ u=0 \text{ on }\partial\Omega. \end{cases}$$



Figure: A membrane with variable height h(x). Figure from Allaire 2004.

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- u(x) is the vertical displacement,
- $A \equiv A(x) = h(x)I$ is the strain tensor, the deformation $\nabla u(x)$ yields a local force of magnitude $A(x)\nabla u(x) \cdot \mathbf{n}(x)$ on the facets of an elementary domain



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- A ≡ A(x) = h(x)I is the strain tensor, the deformation ∇u(x) yields a local force of magnitude A(x)∇u(x) · n(x) on the facets of an elementary domain
- \blacktriangleright f(x) is a distribution of forces applied on the membrane (e.g. a pressure distribution)
- ▶ the membrane is clamped on the boundary of Ω whence u = 0 on $\partial \Omega$

▶ The equation

$$\begin{cases} -\operatorname{div}(A\nabla u) = f \text{ in } \Omega, \\ u = 0 \text{ on } \partial\Omega. \end{cases}$$

is called the **strong** formulation of the model.

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In numerical practice we work with the weak formulation:

Find
$$u \in H^1_0(\Omega)$$
 such that $\int_{\Omega} A(x) \nabla u \cdot \nabla v \mathrm{d}x = \int_{\Omega} f v \mathrm{d}x$ for any $v \in H^1_0(\Omega)$

Let $\Omega \subset \mathbb{R}^d$ a smooth bounded domain. We denote by $H^1(\Omega)$ the Sobolev space

$$H^1(\Omega) = \{ v \mid v \in L^2(\Omega) \text{ and } \nabla v \in L^2(\Omega, \mathbb{R}^d) \}.$$

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We denote by $||v||_{H^1(\Omega)} := ||v||_{L^2(\Omega)} + ||\nabla v||_{L^2(\Omega)}$ the norm on $H^1(\Omega)$, which is a Hilbert space for the inner product

$$(u,v)_{H^1(\Omega)} := \int_{\Omega} (uv + \nabla u \cdot \nabla v) \mathrm{d}x.$$

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Proposition 1

There exists a constant C > 0 such that

 $||\phi||_{L^2(\partial\Omega)} \leq C ||\phi||_{H^1(\Omega)}$ for any $\phi \in \mathcal{C}^{\infty}(\overline{\Omega})$.

Hence the application $\gamma : \phi \mapsto \phi_{|\partial\Omega}$ extends continuously to an application of $\gamma : H^1(\Omega) \to L^2(\partial\Omega)$. $\gamma(v)$ is called the trace of v and is still denoted by $v_{|\partial\Omega}$.

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$$H^1_0(\Omega):=\{v\in H^1(\Omega)\,|\,v_{|\partial\Omega}=0\}.$$

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Proposition 2 (Green formula)

Let $u, v \in H^1(\Omega)$. Then

$$\int_{\Omega} \frac{\partial u}{\partial x_i} \mathrm{d}x = \int_{\partial \Omega} u n_i \mathrm{d}\sigma$$

where $\mathbf{n} = (n_1, \dots, n_d)$ is the outward normal to Ω .

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$$\int_{\Omega} -\operatorname{div}(A\nabla u)v \mathrm{d}x = \int_{\Omega} A\nabla u \cdot \nabla v \mathrm{d}x - \int_{\partial\Omega} v A\nabla u \cdot n \mathrm{d}\sigma = \int_{\Omega} A\nabla u \cdot \nabla v \mathrm{d}x$$

 $\text{ if } v = 0 \text{ on } \partial \Omega.$

▶ The equation

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In numerical practice we work with the weak formulation:

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Proposition 3 (Lax Milgram's theorem)

Let V be a Hilbert space and a : $V \times V \rightarrow \mathbb{R}$ be a coercive symmetric bilinear form:

►
$$a(\lambda u + v, w) = \lambda a(u, w) + a(v, w)$$
 and $a(u, v) = a(v, u)$ for any $u, v, w \in V$, $\lambda \in \mathbb{R}$,

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$$|a(u, v)| \le C||u||_V||v||_V$$
 for any $u, v \in V$ with $C > 0$,

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Then for any $b \in V'$, there exists a unique $u \in V$ solving the variational formulation

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Here,
$$V = H_0^1(\Omega)$$
, $a(u, v) = \int_{\Omega} A(x) \nabla u \cdot \nabla v dx$, $b(v) = \int_{\Omega} f v dx$.

Proposition 4 (Poincaré inequality)

Assume Ω is a smooth bounded domain. There exists C>0 such that

$$\int_{\Omega} |\nabla u|^2 \mathrm{d} x \geqslant C \int_{\Omega} |u|^2 \mathrm{d} x \text{ for any } u \in H^1_0(\Omega).$$

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Proof.

Assume the result is wrong. Then there exists a sequence $(u_n)_{n\in\mathbb{N}}$ such that $\int_{\Omega} |\nabla u_n|^2 dx \leq \frac{1}{n} \int_{\Omega} |u_n|^2 dx.$ Up to a rescaling, we may assume that $\int_{\Omega} |u_n|^2 dx = 1$. Then $||\nabla u_n||_{L^2(\Omega)} \to 0$ while $||u_n||_{H^1(\Omega)}$ remains bounded. By the Rellich theorem, we may assume, up to extracting a subsequence, that $u_n \to u$ strongly in $L^2(\Omega)$ for some $u \in H^1(\Omega)$. Since $||\nabla u_n||_{L^2(\Omega)} \to 0$, the convergence is in fact strong in $H^1(\Omega)$ and we obtain that $\nabla u = 0$, hence u is constant on Ω . Since $u_n \in H_0^1(\Omega)$, we find that u = 0 on Ω , which is not possible because $||u_n||_{L^2(\Omega)} = 1$ implies $||u||_{L^2(\Omega)} = 1$. Corollary:

• if $-\operatorname{div}(A(x)\nabla \cdot)$ is uniformly elliptic, i.e.

$$\exists C > 0 \,, \, A_{ij}(x) \boldsymbol{\xi}_i \boldsymbol{\xi}_j \geqslant C |\boldsymbol{\xi}|^2 \text{ for all } \boldsymbol{\xi} \in \mathbb{R}^d$$

then

$$a(u, u) = \int_{\Omega} A(x) \nabla u \cdot \nabla u \mathrm{d}x \ge C ||\nabla u||_{L^{2}(\Omega)}^{2} \ge C ||u||_{H^{1}(\Omega)}$$

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is a symmetric coercive bilinear form.

Then the weak formulation

Find $u \in H_0^1(\Omega)$ such that $\forall v \in H_0^1(\Omega), a(u, v) = b(v)$,

admits a unique solution $u \in H_0^1(\Omega)$.

The finite element (or Galerkin approximation) method amount to approximate the variational formulation

Find
$$u \in V$$
 such that $\forall v \in V$, $a(u, v) = b(v)$,

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Find
$$u_h \in$$
 such that $\forall v \in V_h$, $a(u_h, v_h) = b(v_h)$, (1)

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Lemma 3

Assume that a satisfies the condition of the Lax-Milgram's theorem. Then eq. (1) admits a unique solution $u_h \in V_h$ which can be obtained by solving a finite-dimensional symmetric linear system.

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Lemma 3

Assume that a satisfies the condition of the Lax-Milgram's theorem. Then eq. (1) admits a unique solution $u_h \in V_h$ which can be obtained by solving a finite-dimensional symmetric linear system.

Proof.

Let
$$(\phi_i)_{1\leq i\leq N_h}$$
 a basis of V_h . Then if $u_h=\sum_{j=1}^{N_h}u_i\phi_i$, eq. (1) is equivalent to

find
$$(u_i)_{1 \leq i \leq N_h}$$
 such that $a\left(\sum_{j=1}^{N_h} u_j \phi_j, \phi_i\right) = b(\phi_i),$

that is $a(\phi_i, \phi_j)u_j = b(\phi_i), \quad 1 \leq i \leq N.$

• The matrix $A_h = (a(\phi_i, \phi_j))_{1 \le i,j \le N_h}$ is called the rigidity matrix.

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- When using the finite element method, it is common to numerically approximate the domain Ω by a triangular or tetrahedral mesh T_h = ∪^N_{i=1}T_i with N triangles/tetrahedra T_i with size h



Figure: Triangular and tetrahedral meshes of a square and a cube.

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- Typically, h is the maximum element size (maximum edge size, or maximum height of a tetrahedron)
- It is common to use $\mathbb{P}_k(\mathcal{T}_h)$ -Lagrange finite elements for the space V_h :

 $\mathbb{P}_k(\mathcal{T}_h) = \{ v \text{ continuous on } \mathcal{T}_h \, | \, v_{\mathcal{T}_i} \text{ is a polynomial of degree less than } k \}.$

Under reasonable assumptions on the mesh T_h it is possible to prove that the variational approximation is convergent:

$$\lim_{h\to 0}||u-u_h||_{H^1(\Omega)}=0.$$

Moreover, if $u \in H_{k+1}(\Omega)$, then we have the error estimate

$$||u - u_h||_{H^1(\Omega)} \le Ch^k ||u||_{H^{k+1}(\Omega)}$$

for some constant C > 0 independent of u and h.

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• there is no need to use higher order finite element if the solution is only in $H^1(\Omega)$.

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is called the strong formulation of the model.

In numerical practice we work with the weak formulation:

$$\mathsf{Find} \ u \in H^1_0(\Omega) \text{ such that } \int_\Omega A(x) \nabla u \cdot \nabla v \mathrm{d} x = \int_\Omega f v \mathrm{d} x \text{ for any } v \in H^1_0(\Omega)$$

• Regularity estimates show that if $f \in H^k(\Omega)$, then $u \in H^{k+2}(\Omega)$.

Numerical solution in FreeFEM:



Figure: Numerical solution of the membrane problem with uniform A = I and f = -1 on an annulus domain Ω .

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- ▶ a design variable and an **admissible set** : thickness h(x) of the membrane, shape of the domain Ω

$$J(u) := \int_{\Omega} f u \mathrm{d} x = \int_{\Omega} A \nabla u \cdot \nabla u \mathrm{d} x = \int_{\Omega} h(x) |\nabla u|^2 \mathrm{d} x.$$

A common objective function: the compliance b(u):

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Minimum and maximum thickness if the design variable is h:

$$\forall x \in \Omega, h_{\min} \leq h(x) \leq h_{\max(x)}$$

In that case, we need to deal with a **point-wise** constraint.

For instance, an optimal design problem of the membrane reads

$$\min_{\Omega} \quad J(u_{\Omega}) := \int_{\Omega} f u dx$$
$$s.t. \left\{ \operatorname{Vol}(\Omega) := \int_{\Omega} dx = \operatorname{Vol}_{\Omega} dx \right\}$$

and $u_{\Omega} \in H^1_0(\Omega)$ is solution to $a(u_{\Omega}, v) = b(v)$ for all $v \in H^1_0(\Omega)$.

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- formulating an optimal design problem: modelling of the performance through an objective function, and of the specification constraints.

- 1. The membrane's equation, reminders on FEM and weak formulations
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Linear elasticity



$$\begin{cases} -\operatorname{div}(\sigma_s(\boldsymbol{u})) = \boldsymbol{f} \text{ in } \Omega \\ \boldsymbol{u} = 0 \text{ on } \Gamma_D \\ \sigma_s(\boldsymbol{u}) \cdot \boldsymbol{n} = \boldsymbol{g} \text{ on } \Gamma_N \\ \sigma_s(\boldsymbol{u}) \cdot \boldsymbol{n} = 0 \text{ on } \Gamma \end{cases}$$

Figure: A cantilever beam subjected to traction forces on Γ_N and zero displacement on Γ_D . Figure from Allaire, 2004.
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 $\sigma_s(u) := Ae(u) \in \mathbb{R}^{d \times d}$ is the strain or solid stress tensor. The Hooke's law states that

$$Ae(u) = 2\mu e(u) + \lambda \operatorname{Tr}(e(u))I$$
 with $e(u) := \frac{\nabla u + \nabla u^T}{2}$.

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 μ and λ are the Lamé coefficients, related to the Young modulus and Poisson ratio from the formula:

$$\lambda = \frac{\nu E}{(1+\nu)(1-(d-1)\nu)}, \quad \mu = \frac{E}{2(1+\nu)}.$$

The weak formulation of

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is find $\boldsymbol{u} \in H^1_0(\Omega, \mathbb{R}^d)$ such that for any $\boldsymbol{v} \in H^1_0(\Omega, \mathbb{R}^d)$,

$$\int_{\Omega} Ae(\boldsymbol{u}) : e(\boldsymbol{v}) dx = \int_{\Omega} \boldsymbol{f} \cdot \boldsymbol{v} dx + \int_{\Gamma_N} \boldsymbol{g} \cdot \boldsymbol{v} d\sigma,$$

where

$$H^1_0(\Omega, \mathbb{R}^d) = \{ \mathbf{v} \in H^1(\Omega, \mathbb{R}^d) \mid \mathbf{v} = 0 \text{ on } \Gamma_D \}.$$

The compliance minimization problem reads

$$\min_{\Omega \subset D} \quad J(\Omega, \boldsymbol{u}(\Omega)) := \int_{\Omega} \boldsymbol{f} \cdot \boldsymbol{u} dx + \int_{\Gamma_N} \boldsymbol{g} \cdot \boldsymbol{v} d\sigma = \int_{\Omega} A\boldsymbol{e}(\boldsymbol{u}) : \boldsymbol{e}(\boldsymbol{u}) dx$$

s.t. $\operatorname{Vol}(\Omega) := \int_{\Omega} dx = V_{target}.$



Figure: Setting of the cantilever optimization problem.

▶ Ideally, the industry seeks to consider rather the mass minimization problem

$$\begin{split} \min_{\Omega\subset D} & \operatorname{Vol}(\Omega) \\ s.t. \; e(\boldsymbol{u}): e(\boldsymbol{u})(x) \leq \operatorname{VM}_0 \text{ for any } x\in\Omega, \end{split}$$

where the constraint imposes a Von-Mises upper bound on the strain energy to prevent premature fatigue. This point-wise stress constraint is delicate to implement.

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where the constraint imposes a Von-Mises upper bound on the strain energy to prevent premature fatigue. This point-wise stress constraint is delicate to implement.

 Non-linear models or more complex constitutive laws can be considered to account for plasticity or more realistic phenomena.

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v is the fluid velocity field, *p* is the (static) pressure, *ρ* is the fluid density.
 σ_f(*v*, *p*) is the fluid stress tensor. For Newtonian fluids, we have

$$\sigma_f(\mathbf{v},\mathbf{p}) = 2\nu e(\mathbf{v}) - pI, \quad e(\mathbf{v}) = \frac{\nabla \mathbf{v} + \nabla \mathbf{v}^T}{2}$$

where ν is the (dynamic) fluid viscosity, so that $-\operatorname{div}(\sigma_f(\mathbf{v}, \mathbf{p})) = -\nu \Delta \mathbf{v} + \nabla \mathbf{p}$.



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$$\nabla \mathbf{v} \, \mathbf{v} = \left(\sum_{j=1}^d \partial_j \mathbf{v}_i \mathbf{v}_j \right)_{1 \le i \le d};$$



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Remark: the time-derivative $\partial_t \mathbf{v}$ can be delicate to handle in an optimal design problem:

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We may assume in many practical cases that $\partial_t \mathbf{v} = 0$, \mathbf{v} does not depend on time.

• The Reynolds number is the ratio between convection $\rho \nabla v v$ and diffusion $\nu \Delta v$. It can be estimated as

$$\operatorname{Re} := \frac{\rho L||v_0||}{\nu}$$

where L is a characteristic size of the problem (e.g. the width of the inlet).

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- ▶ There is no universal definition because *L* can be chosen differently !
- ▶ However, when Re is large (typically $\text{Re} \ge 10^3$), convection effects dominate and the numerical simulation maybe very difficult to solve due to **turbulence** and **boundary layer effects**



Figure: Example of turbulent flows featuring vortices. Images from Wikipedia and https://studiousguy.com/turbulent-flow-examples/

lt is difficult to resolve the Navier-Stokes equations when Re starts to be greater than 10^3 . The best large scale simulations do not go much larger than Re = 6000.



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- \blacktriangleright In the aeronautic industry, a flight encounters Reynolds number of the order of 10^6 .
- A variety of turbulence models (Large Eddy Simulation (LES), Reynold Average Navier Stokes (RANS)...) are commonly used in the aeronautic industry to obtain approximate simulation of turbulent flows.

For our applications, we will assume moderate Reynolds number ($\text{Re} \leq 500$). The steady-state Navier-Stokes equations are nonlinear and requires a different numerical treatment:

the nonlinearity can be solved using the Newton method. (ν, p) are solutions to the following variational problem: find (ν, p) ∈ ν₀ + V_{ν,p}(Γ) such that

$$\forall (\boldsymbol{w}, \boldsymbol{q}) \in V_{\boldsymbol{v}, \boldsymbol{p}}(\Gamma) \quad \int_{\Omega_f} \left[\sigma_f(\boldsymbol{v}, \boldsymbol{p}) : \nabla \boldsymbol{w} + \rho \boldsymbol{w} \cdot \nabla \boldsymbol{v} \cdot \boldsymbol{v} - \boldsymbol{q} \mathrm{div}(\boldsymbol{v}) \right] \mathrm{dx} = \int_{\Omega_f} \boldsymbol{f}_f \cdot \boldsymbol{w} \mathrm{dx};$$

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$$\int_{\Omega_f} [\sigma_f(\delta \mathbf{v}_k, \delta p_k) : \nabla \mathbf{w} + \rho \mathbf{w} \cdot \nabla \mathbf{v}_k \cdot \delta \mathbf{v}_k + \rho \mathbf{w} \cdot \nabla (\delta \mathbf{v}_k) \cdot \mathbf{v}_k - q \operatorname{div}(\delta \mathbf{v}_k) - \delta p_k \operatorname{div}(\mathbf{w})] \, \mathrm{dx}$$

$$= \int_{\Omega_f} \mathbf{f}_f \cdot \mathbf{w} \mathrm{dx}.$$

For our applications, we will assume moderate Reynolds number ($\text{Re} \leq 500$). The steady-state Navier-Stokes equations are nonlinear and requires a different numerical treatment:

the nonlinearity can be solved using the Newton method. (ν, p) are solutions to the following variational problem: find (ν, p) ∈ ν₀ + V_{ν,p}(Γ) such that

$$\forall (\boldsymbol{w}, \boldsymbol{q}) \in V_{\boldsymbol{v}, \boldsymbol{p}}(\Gamma) \quad \int_{\Omega_f} \left[\sigma_f(\boldsymbol{v}, \boldsymbol{p}) : \nabla \boldsymbol{w} + \rho \boldsymbol{w} \cdot \nabla \boldsymbol{v} \cdot \boldsymbol{v} - \boldsymbol{q} \mathrm{div}(\boldsymbol{v}) \right] \mathrm{dx} = \int_{\Omega_f} \boldsymbol{f}_f \cdot \boldsymbol{w} \mathrm{dx};$$

where $V_{\boldsymbol{v},\rho}(\Gamma) = \{(\boldsymbol{w},q) \in H^1(\Omega_f,\mathbb{R}^d) \times L^2(\Omega_f)/\mathbb{R} \mid \boldsymbol{w} = 0 \text{ on } \partial\Omega_f \}.$

At each Newton step, an increment (δν_k, δp_k) is computed by solving the linearized Navier-Stokes equations

Find
$$(\delta \mathbf{v}_k, \delta \mathbf{p}_k) \in V_{\mathbf{v}, \rho}(\Gamma)$$
 such that $\forall (\mathbf{w}, q) \in V_{\mathbf{v}, \rho}(\Gamma)$,

$$\int_{\Omega_f} [\sigma_f(\delta \mathbf{v}_k, \delta \mathbf{p}_k) : \nabla \mathbf{w} + \rho \mathbf{w} \cdot \nabla \mathbf{v}_k \cdot \delta \mathbf{v}_k + \rho \mathbf{w} \cdot \nabla (\delta \mathbf{v}_k) \cdot \mathbf{v}_k - q \operatorname{div}(\delta \mathbf{v}_k) - \delta \mathbf{p}_k \operatorname{div}(\mathbf{w})] \, \mathrm{d}x$$

$$= \int_{\Omega_f} \mathbf{f}_f \cdot \mathbf{w} \mathrm{d}x.$$

The next iterate
$$(\mathbf{v}_{k+1}, \mathbf{p}_{k+1})$$
 is then obtained by setting $\mathbf{v}_{k+1} := \mathbf{v}_k + \delta \mathbf{v}_k, \quad p_{k+1} := p_k + \delta p_k.$

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- Existence and uniqueness of a solution to the steady-state NS equation is guaranteed for low Reynolds number (by a perturbation argument of the Stokes system).
- For large Reynolds number, there may exist several solutions to the NS equations (due to nonlinearity). The relevance of the physical modelling can then be questioned because the trajectories of the solutions generally converge to a limit cycle (or worse, to a global attractor) and not to a steady state.

Common design functionals:

▶ The **drag** or sum of the friction forces:

$$\mathrm{Drag}(\boldsymbol{\Gamma},\boldsymbol{\nu}(\boldsymbol{\Gamma}),\boldsymbol{p}(\boldsymbol{\Gamma})):=\int_{\Omega_f}\sigma_f(\boldsymbol{\nu},\boldsymbol{p}):\nabla\boldsymbol{\nu}\mathrm{d}x=\int_{\Omega_f}2\nu\boldsymbol{e}(\boldsymbol{\nu}):\boldsymbol{e}(\boldsymbol{\nu})\mathrm{d}x.$$

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(It is the equivalent of the compliance for fluids)

▶ The lift or forces generated by the flow around an obstacle along one direction:

$$\operatorname{Lift}(\Gamma, \boldsymbol{\nu}(\Gamma), \boldsymbol{p}(\Gamma)) := -\int_{\Gamma} \boldsymbol{e}_{\boldsymbol{y}} \cdot \sigma_f(\boldsymbol{\nu}, \boldsymbol{p}) \cdot \boldsymbol{n} \mathrm{d}\boldsymbol{s},$$

Typical aerodynamic design problem:



Figure: Setting of the aerodynamic design problem.

$$\begin{array}{ll} \min & -\operatorname{Lift}({\boldsymbol{\Gamma}}, {\boldsymbol{\nu}}({\boldsymbol{\Gamma}}), p({\boldsymbol{\Gamma}})) \\ & \\ \text{S.t.} \begin{cases} & \operatorname{Drag}({\boldsymbol{\Gamma}}, {\boldsymbol{\nu}}({\boldsymbol{\Gamma}}), p({\boldsymbol{\Gamma}})) \leq \operatorname{DRAG}_0 \\ & & \operatorname{Vol}(\Omega_f) = V_0 \\ & \\ \boldsymbol{\mathcal{X}}(\Omega_s) := \frac{1}{|\Omega_s|} \int_{\Omega_s} x \mathrm{d}x = \boldsymbol{x}_0. \end{array} \end{cases}$$
Remarks:

the functional Drag(Γ, ν(Γ), ρ(Γ)) is also a measure analogous to the (static) pressure drop:

$$extsf{DP}(\Omega_f) := \int_{\partial \Omega_{f, extsf{out}}} p \mathrm{d}\sigma - \int_{\partial \Omega_{f, extsf{in}}} p \mathrm{d}\sigma.$$

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- ► this quantity is not numerically well-behaved because the pressure belongs to $L^2(\Omega_f)$ and may not have in general a well defined trace numerically.
- This problem also arises for the Lift functional, but can be bypassed by resorting to an equivalent volume formulation.

$$\operatorname{Lift}(\Gamma) = \int_{\Omega_f} (\mathcal{X} \mathbf{f}_f \cdot \mathbf{e}_y - \rho \mathcal{X} \mathbf{e}_y \cdot \nabla \mathbf{v} \cdot \mathbf{v} - \nabla \mathcal{X} \cdot \sigma_f(\mathbf{v}, \mathbf{p}) \cdot \mathbf{e}_y) \mathrm{d}x$$

where \mathcal{X} is a scalar field satisfying $\mathcal{X} = 1$ on $\partial \Omega_s$.

- 1. The membrane's equation, reminders on FEM and weak formulations
- 2. Linear elasticity
- 3. Fluid flows
- 4. Heat diffusion
- 5. Coupled physics:
 - 5.1 convective heat transfer 5.2 fluid-structure interaction



Figure: A bi-material distrubution of two conductive media with conductivity k_s and k_v .

• k_f , k_s : conductivity coefficients in Ω_f , Ω_s ;

$$\begin{aligned} -\operatorname{div}(k_f \nabla T_f) &= Q_f & \text{in } \Omega_f \\ -\operatorname{div}(k_s \nabla T_s) &= Q_s & \text{in } \Omega_s \\ T &= T_0 & \text{on } \partial \Omega_T^D \\ -k_f \frac{\partial T_f}{\partial \mathbf{n}} &= h & \text{on } \partial \Omega_T^N \cap \partial \Omega_f \\ -k_s \frac{\partial T_s}{\partial \mathbf{n}} &= h & \text{on } \partial \Omega_T^N \cap \partial \Omega_s \\ T_f &= T_s & \text{on } \Gamma \\ -k_f \frac{\partial T_f}{\partial \mathbf{n}} &= -k_s \frac{\partial T_s}{\partial \mathbf{n}} & \text{on } \Gamma, \end{aligned}$$



Figure: A bi-material distrubution of two conductive media with conductivity k_s and k_v .

- ► k_f , k_s : conductivity coefficients in Ω_f , Ω_s ;
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- \blacktriangleright k_f , k_s : conductivity coefficients in Ω_f , Ω_s ;
- \triangleright Q_f , Q_s : volume heat sources
- \blacktriangleright T_0 : prescribed temperature on isothermal boundary $\partial \Omega_{\tau}^D$.
- ▶ *h*: heat loss on the boundary $\partial \Omega_T^N$. If h = 0, we say that the boundary is adiabatic.

The variational formulation reads find $T \in T_0 + V_T(\Gamma)$ such that, for any $S \in V_T(\Gamma)$,

$$\int_{\Omega_s} k_s \nabla T \cdot \nabla S \mathrm{d}x + \int_{\Omega_f} k_f \nabla T \cdot \nabla S \mathrm{d}x = \int_{\Omega_s} Q_s S \mathrm{d}x + \int_{\Omega_f} Q_f S \mathrm{d}x + \int_{\partial \Omega_T^N} h S \mathrm{d}s.$$

where

$$V_{\mathcal{T}}(\Gamma) = \{ S \in H^1(D) \mid S = 0 \text{ on } \partial \Omega^D_{\mathcal{T}} \},$$

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Not harder to solve than a Laplacian.

Minimizing the average temperature with a cooling material Ω_f :



 $\min_{\Gamma} \quad J(\Gamma, T(\Gamma)) = \int_{D} T dx$ s.t. $\operatorname{Vol}(\Omega_{f}) \leq V_{target}.$

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5. Coupled physics:

- 5.1 convective heat transfer
- 5.2 fluid-structure interactions
- 5.3 thermoelasticity

The heat density carried out through convective transport by an incompressible flow v is given by

$$-\rho c_{\rho} \operatorname{div}(\mathbf{v} T_f) = -\rho c_{\rho} \mathbf{v} \cdot \nabla T_f$$

where ρ is the fluid density and c_p the heat capacity of the fluid.

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Once v is determined by the Navier-Stokes equations...

$$\begin{cases} -\operatorname{div}(\sigma_f(\boldsymbol{v},\boldsymbol{p})) + \rho \nabla \boldsymbol{v} \, \boldsymbol{v} = \boldsymbol{f}_f & \text{in } \Omega_f \\ \operatorname{div}(\boldsymbol{v}) = \boldsymbol{0} & \text{in } \Omega_f \\ \boldsymbol{v} = \boldsymbol{v}_0 & \text{on } \partial \Omega_f^D \\ \sigma_f(\boldsymbol{v},\boldsymbol{p})\boldsymbol{n} = \boldsymbol{0} & \text{on } \partial \Omega_f^N \\ \boldsymbol{v} = \boldsymbol{0} & \text{on } \Gamma \end{cases}$$

then one can solve the advection-diffusion equation

$$\begin{split} -\operatorname{div}(k_{f}\nabla T_{f}) + \rho c_{\rho} \mathbf{v} \cdot \nabla T_{f} &= Q_{f} & \text{in } \Omega_{f} \\ -\operatorname{div}(k_{s}\nabla T_{s}) &= Q_{s} & \text{in } \Omega_{s} \\ T &= T_{0} & \text{on } \partial \Omega_{T}^{D} \\ -k_{f} \frac{\partial T_{f}}{\partial \mathbf{n}} &= h & \text{on } \partial \Omega_{T}^{N} \cap \partial \Omega_{f} \\ -k_{s} \frac{\partial T_{s}}{\partial \mathbf{n}} &= h & \text{on } \partial \Omega_{T}^{N} \cap \partial \Omega_{s} \\ T_{f} &= T_{s} & \text{on } \Gamma \\ -k_{f} \frac{\partial T_{f}}{\partial \mathbf{n}} &= -k_{s} \frac{\partial T_{s}}{\partial \mathbf{n}} & \text{on } \Gamma, \end{split}$$



An optimal heat transfer test case:



Figure: Setting of a convective heat transfer test case.

$$\begin{split} \min_{\Gamma} & J(\Gamma, \boldsymbol{v}(\Gamma), T(\Gamma)) := -\int_{\Omega_f} \rho c_{\rho} \boldsymbol{v} \cdot \nabla T \, \mathrm{d}x \\ s.t. \begin{cases} \mathsf{DP}(\rho(\Gamma)) := \int_{\partial \Omega_f^D} p \, \mathrm{d}s - \int_{\partial \Omega_f^N} p \, \mathrm{d}s \leq \mathsf{DP}_{static} \\ \mathrm{Vol}(\Omega_f) = V_{target}. \end{cases} \end{split}$$

$$\begin{split} \min_{\Gamma} \quad & J(\Gamma, \boldsymbol{v}(\Gamma), T(\Gamma)) := -\int_{\Omega_f} \rho c_p \boldsymbol{v} \cdot \nabla T \, \mathrm{d} \boldsymbol{x} \\ & s.t. \begin{cases} \mathrm{DP}(\boldsymbol{p}(\Gamma)) := \int_{\partial \Omega_f^D} \boldsymbol{p} \, \mathrm{d} \boldsymbol{s} - \int_{\partial \Omega_f^N} \boldsymbol{p} \, \mathrm{d} \boldsymbol{s} \leq \mathrm{DP}_{static} \\ & \mathrm{Vol}(\Omega_f) = V_{target}. \end{split}$$

$$\min_{\Gamma} \quad J(\Gamma, \boldsymbol{\nu}(\Gamma), T(\Gamma)) := -\int_{\Omega_f} \rho c_p \boldsymbol{\nu} \cdot \nabla T dx \\ s.t. \begin{cases} \mathsf{DP}(\boldsymbol{\rho}(\Gamma)) := \int_{\partial \Omega_f^D} p ds - \int_{\partial \Omega_f^N} p ds \leq \mathsf{DP}_{static} \\ \operatorname{Vol}(\Omega_f) = V_{target}. \end{cases}$$

 $J(\Gamma, \mathbf{v}(\Gamma), T(\Gamma)) := -\int_{\Omega_f} \rho c_p \mathbf{v} \cdot \nabla T dx$ is the opposite of the heat transferred from the inlet to the outlet, indeed:

$$\int_{\Omega_f} \rho c_p \boldsymbol{v} \cdot \nabla T d\boldsymbol{x} = \int_{\partial \Omega_f} \rho c_p T(\boldsymbol{v} \cdot \boldsymbol{n}) d\boldsymbol{s} = \int_{\partial \Omega_f^N} \rho c_p T(\boldsymbol{v} \cdot \boldsymbol{n}) d\boldsymbol{s} + \int_{\partial \Omega_f^D} \rho c_p T_0(\boldsymbol{v}_0 \cdot \boldsymbol{n}) d\boldsymbol{s}.$$



▶ v = 0 on the deformed solid interface: v(x + u(x)) = 0 for $x \in \Gamma$;



- ▶ $\mathbf{v} = 0$ on the deformed solid interface: $\mathbf{v}(x + \mathbf{u}(x)) = 0$ for $x \in \Gamma$;
- Equality of the normal stresses on the deformed solid interface: $\sigma_s(u)\mathbf{n} = Ae(u) \cdot \mathbf{n} = \sigma_f(\mathbf{v}, p) \cdot \mathbf{n}$ on $x \in (l + u)(\Gamma)$.



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However, the equations of linear elasticity assume that u is small. Therefore a convenient, straightforward approximation is

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The fluid-structure interaction can be modelled by an extra-Neumann boundary condition on the solid interface.

The model is weakly coupled.

One can still maximize the rigidity of the solid structure, which is forced by the incoming flow: $\hat{}$

$$\min_{\Gamma} \quad J(\Gamma, \boldsymbol{u}(\Gamma)) = \int_{\Omega_s} Ae(\boldsymbol{u}) : e(\boldsymbol{u}) dx$$

s.t. $\operatorname{Vol}(\Omega_s) = V_{target}.$

One can still maximize the rigidity of the solid structure, which is forced by the incoming flow: $\hat{}$

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s.t. $\operatorname{Vol}(\Omega_s) = V_{target}.$

The variable $u(\Gamma)$ depends implicitly on $v(\Gamma)$!

Thermal dilation due to an elevated temperature T can be modelled by an extra term in the Hooke's law:

$$\sigma_s(oldsymbol{u}, T_s) := Ae(oldsymbol{u}) - lpha(T_s - T_{ ext{ref}})I$$
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Thermal dilation due to an elevated temperature T can be modelled by an extra term in the Hooke's law:

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▶ $\alpha > 0$ is called the thermal expansion coefficient and T_{ref} is the temperature at rest.

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Thermoelasticity can be coupled to the convection-diffusion model which determines the temperature field T_s





• Incompressible Navier-Stokes equations for velocity and pressure (\mathbf{v}, p) in Ω_f

$$-\operatorname{div}(\sigma_f(\boldsymbol{v},\boldsymbol{p})) + \rho \nabla \boldsymbol{v} \, \boldsymbol{v} = \boldsymbol{f}_f \text{ in } \Omega_f$$



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• Convection-diffusion for the temperature T in Ω_f and in Ω_s :

$$-\operatorname{div}(k_f \nabla T_f) + \rho \mathbf{v} \cdot \nabla T_f = Q_f \quad \text{in } \Omega_f$$
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▶ Incompressible Navier-Stokes equations for velocity and pressure (v, p) in Ω_f

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Linear elasticity with fluid-structure interaction for mechanical deformation u in Ω_s :

$$-\operatorname{div}(\sigma_s(\boldsymbol{u}, T_s)) = \boldsymbol{f}_s \quad \text{in } \Omega_s$$
$$\sigma_s(\boldsymbol{u}, T_s) \cdot \boldsymbol{n} = \sigma_f(\boldsymbol{v}, \boldsymbol{p}) \cdot \boldsymbol{n} \quad \text{on } \Gamma.$$

Exercise

Implement one of the physical models of your choice in FreeFEM: write a PDE solver that computes u, T, or (v, p) for a simple domain geometry.

- https://doc.freefem.org
- https://modules.freefem.org/

earning by Examples	Mathematical Models	
Documentation	Summary.	
Anguage references Mathematical Models Static problems	This chapter goes deeper into a number of problem which was only an introduction.	ts that FreeFEM can solve. It is a complement to the Tutorial pr
Elasticity	Users are invited to contribute to make this models	database grow.
Non-linear static problems	O Previous topic: External libraries	Next topic: Static problems
Eigen value problems		
Evolution problems		
Navier-Stokes equations		
Variational Inequality		
Domain decomposition		
Fluid-structure coupled problem		
Fluid-structure coupled problem Transmission problem		
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